C90B WEIGHT & BALANCE/PERFORMANCE

V _{MC}	Vso	V1 / VR / V2	Vy	VYSE	VA	VF	Vмо	
80	78	Computed	112	108	169	Aprch: 184	Full: 140	226

VREF		Vlo		VGLIDE	Emer. Descent	Sustained Ice
Flaps up	Flaps 100%	Up	Down	125	182	≥140
115	101	163	182			

WEIGHT AND BALANCE

ITEM	WEIGHT		MOMENT / 100
Basic Empty Weight (N370U)	6845.27		10332.93
Pilot and Co-Pilot			
Passengers—FWD Club Seats			
Passengers—AFT Club Seats			
PAX—Aisle Facing Storage Seat			
Passenger—Lavatory Seat			
Rear Baggage Compartment			
FWD Cabinet			
AFT Cabinet			
Equals Zero Fuel Weight			
*Determine Max T/off weight to achi	eve Positive Single E	ngine Climb @ Lift	-off = Lbs.
Fuel [384 gallons Max. Usable]			
Equals Ramp Weight (10,160 lbs)			
(Start / Taxi Fuel Burn-off)	-60.0		-93
Equals Take-off Weight			
(Fuel Consumed in Flight)	-		XXXXXXXXXXXXXXXXXXX
Total Fuel Remaining			
Zero Fuel Weight	+	-	+
Equals Landing Weight		—	
Maximum Take-off Weight: 10,100 lbs. →	□Forward C.G. Limit: 14	45.0 Aft C.G. Limit:	160.0

*Verify that both Take off and Landing Weights and Moments are Within Limits (Use POH)

Surface Weather					
1pute					

Additional Weather	
6000'	6000'
9000'	9000'
12000'	12000'
18000'	18000'
24000'	24000'
Interpolate for Cruise Alt.	

ISA conversion @ Cruise

 Surface Weather @ Destination

 Wind

 Visibility

 Sky Condition

 Temperature

 Altimeter

Fuel = 6.7 lbs. / Gal.

Accelerated Stop Distance	
Accelerated Go Distance	
Takeoff Distance	
V_1 / V_R Speed	
M.E. Climb Gradient/V2	
Rate of Climb Two Engines	
S.E. Climb Gradient	
Rate of Climb Single Engine	
S.E. Absolute Ceiling	
S.E. Climb Gradient Rate of Climb Single Engine S.E. Absolute Ceiling	

PERFORMANCE

Single engine Service Ceiling							
Rate of Climb @ TPA							
To Climb			Fuel:	Dist:			
Cruise Power (Select POWER or RANGE)							
Torque:	Fuel/Lbs/Hr:			TAS	TAS:		
One Engine Inoperative Max. Cruise Power							
Torque: Fuel/Lb					5:		
To Descend			Fuel:		Dist:		
Landing Distance							